Advanced Ceramic Materials for Future Aerospace Applications

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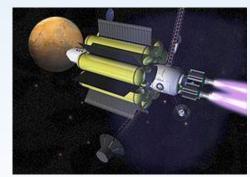
Widespread Use of Ceramics in Multiple Aerospace Systems













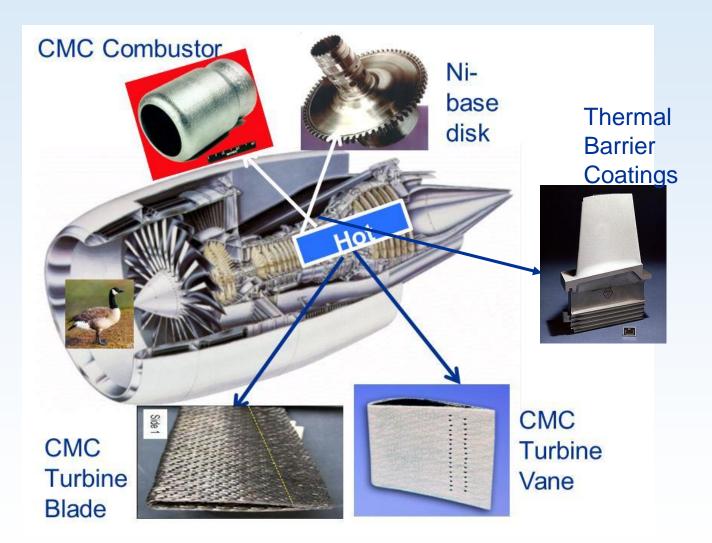






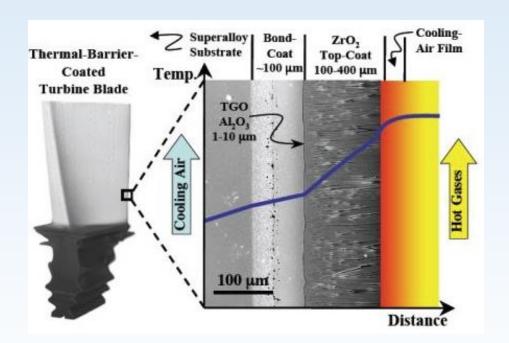


Ceramic Materials in Gas Turbine Engines





Thermal Barrier Coatings

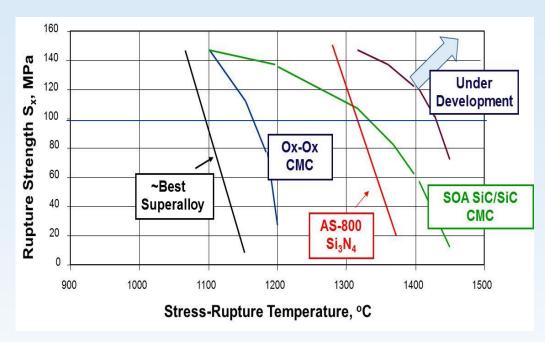


Future challenges:

- Increased temperature capability
- Low thermal conductivity
- Erosion resistance
- Resistance to molten sand/glass deposit



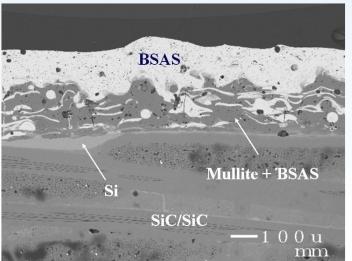
CMCs for Gas Turbine Engine Hot Section



SiC/SiC CMC preferred

Environmental Barrier Coatings Required for CMCs EBC - Bondcoat-

CMC



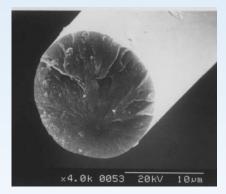


Challenges for Increasing Temperature Capability of SiC/SiC CMCs for Gas Turbine Engines

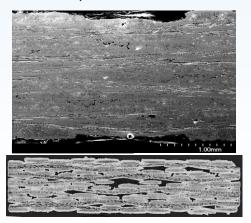
2400°F Today

2700°F + Future

Advanced SiC Fiber



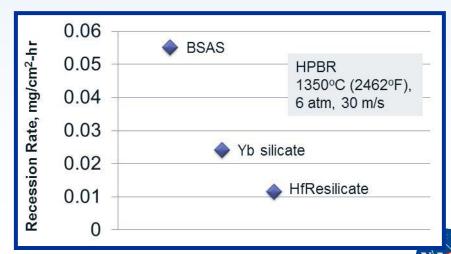
Dense, Si-free Matrix



Durable Environmental Barrier Coatings with 2700°F+ Capability







Increasing Use of CMCs in Aircraft

Boeing – CMC Exhaust Nozzle





NASA
Environmentally
Responsible
Aviation Project
– CMC Nozzle
Demonstration





GE Passport
Engine Exhaust
Nozzle



Ceramic Matrix Composites for Hypersonic Vehicles

3000 F + temperature capability required

Benefit: Reduced weight



Leading Edges

Hypersonic Control

Surfaces

Reentry TPS

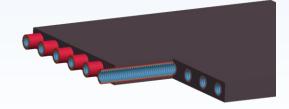
Leading Edges

> Exhaust-Washed Structure





Acreage TPS



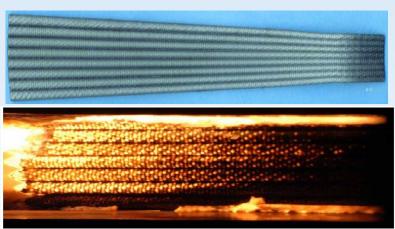
Actively Cooled CMC Combustor



Control Surfaces



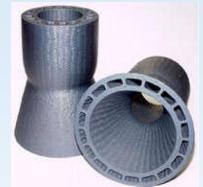
Cooled Ceramic Matrix Composite Structures in Hypersonic and Rocket Propulsion



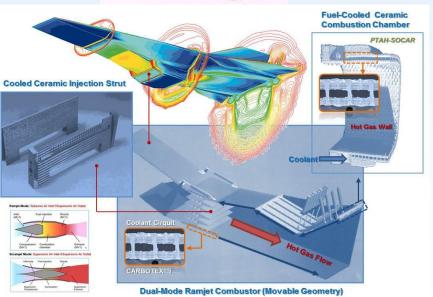
NASA GRC - Teledyne



AFRI



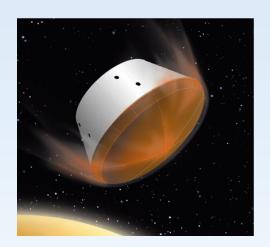
NASA GRC



EADS - Astrium



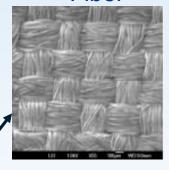
High Temperature Materials for Planetary Entry, Descent, and landing (EDL)



Thermal Barrier Seals



Woven SiC Fiber



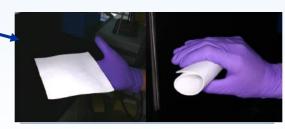




Outer Fabric

Aerogel Insulation

Gas Barrier



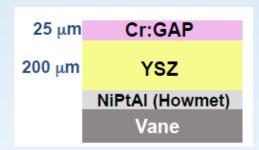
High Temperature Ceramic Aerogel

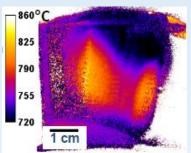
High Temperature Thin Film Ceramic Sensors

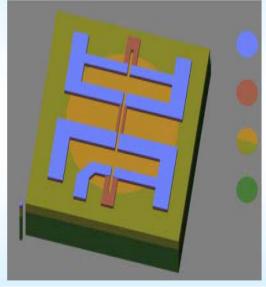
SiC Pressure Sensor











metal contacts Ti / TaSi₂ / Pt

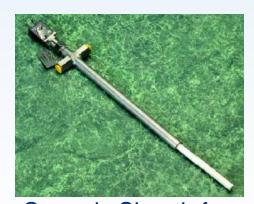
strain gages n-type SiC

isolation layer p-type SiC

substrate n-type SiC



Multifunctional TaN-Based Sensors



Ceramic Sheath for 2400°C – Capable Temperature Probe

Ion and Hall Thrusters for In-Space Propulsion









BN ceramic discharge chamber – sputter erosion limits life



 Provides higher exhaust velocity than chemical rockets – reduces propellant mass and reduction in launch mass

Life Limiting Mechanisms:

- Ion sputter erosion of electrodes and ceramics
- Erosion and depletion of cathode materials

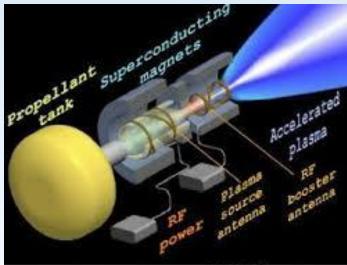
Material Needs:

- High temperature sputter resistant electrodes and ceramics
- Long-life, low work function cathode (LaB₆ ZrB₂ eutectic promising)



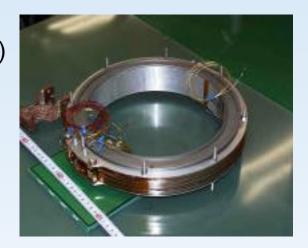
Superconducting Ceramics in Electric Propulsion

Variable Specific Impulse Magnetoplasma Rocket (VASIMR)



Schematic overview of the VASIMR® system

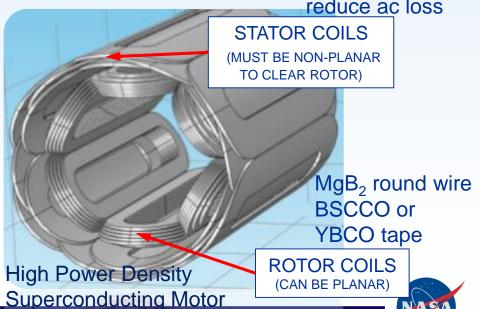




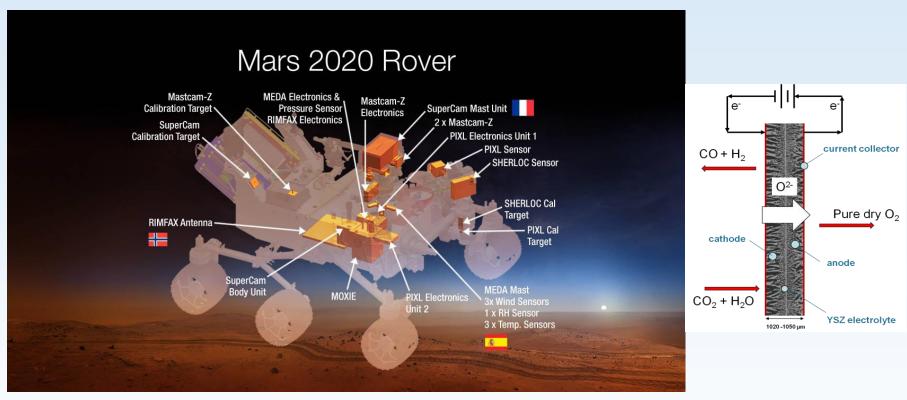
Superconducting magnet for VASIMR

> MgB₂ round wire - Small diameter to

reduce ac loss



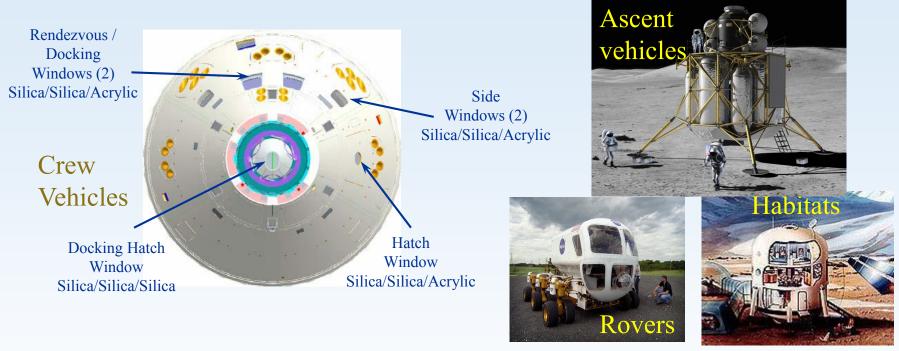
Solid Oxide Electrolyzer for Oxygen Generation on Mars

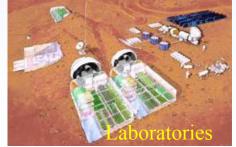


Mars Oxygen ISRU Experiment (MOXIE)

- Extract oxygen from the horrible Martian atmosphere by breaking down carbon dioxide.
- Enable a manned Mars mission to have oxygen ready and waiting when they arrived by sending remote oxygen generators to the surface ahead of time.

Glass Windows in Space Systems









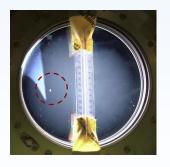


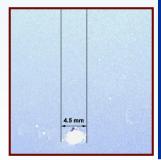
Advanced Window Glass Materials for Space Systems

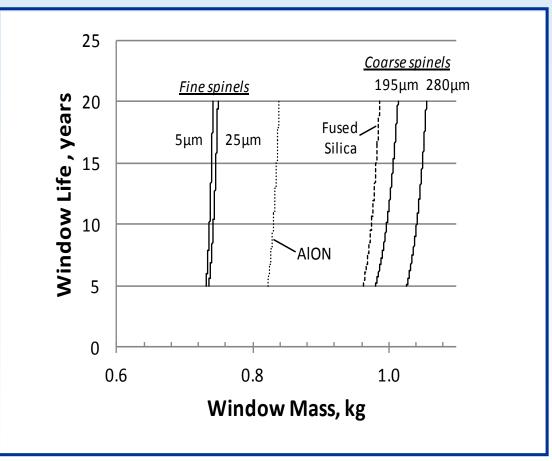
Damage of Glass
Windows due to
Micrometeroid Impact



Damaged Space Shuttle window



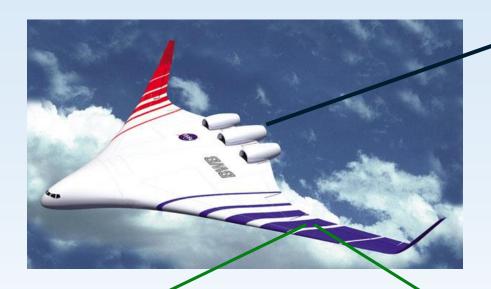




Damaged ISS window



Application of Piezoceramic Materials



ENGINES

Piezoeletric Devices

- Energy harvesting
- Power amplification
- Vibration suppression
- Noise suppression

AIRFRAME

Piezoresistive Devices

- Embedded pressure sensors
- Embedded strain sensors

Piezoelectric Devices

- Energy harvesting
- Cabin noise suppression
- Active flow control
- Variable control surfaces

Challenges:

- High temperature capability (>> 300°C)
- Large displacement
- Integration with structure and durability of integrated structure
- Multifunctionality



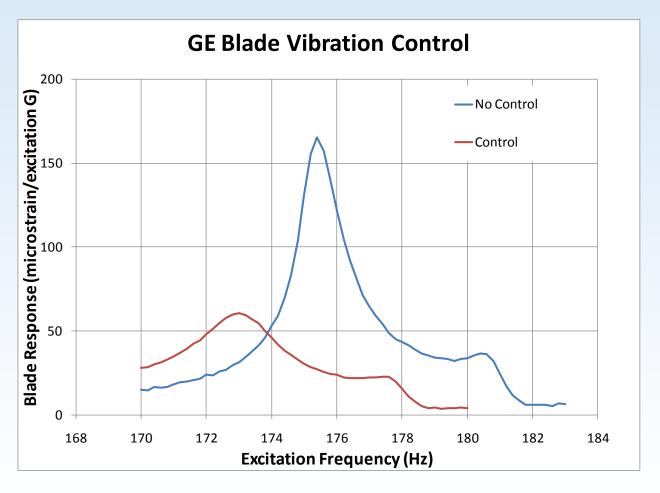
Piezoceramic Patches for Controlling Vibration of PMC Fan Blades



Fan Blade with Piezo patches



Fan Blade with Piezo Patch in Test Rig





Demonstration of Smart Rotor for Helicopters Using Piezoceramic Materials

- Smart rotor incorporates cutting edge changes to MD900 baseline rotor
 - Trailing edge control flap
 - Piezo-electric "smart" material actuators
- Effectiveness of flap for noise and vibration control demonstrated
- Closed-loop feedback control applied for first time to full-scale active rotor
- Initial demonstration of blade displacement technique





Power Conversion and Energy Storage System

Hybrid Electric Long-Aircraft Durati

Long-Duration EVA





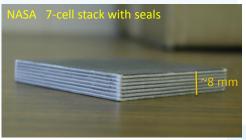




- Need 2 4X increase in energy density of batteries
- Need > 5X increase in power density of fuel cell for electric aircraft

High Power Density Solid Oxide Fuel Cell





SiC Power
Electronics for
High Power Density
and Radiation
Tolerant Power
Processing System

High Energy Density Batteries



Ceramic electrolyte for solid state batteries

Ceramic cathode

Standard Power Module

All oxide ceramic components

Multifunctional systems with structural load bearing capability ??



Ceramics in Satellite Communication



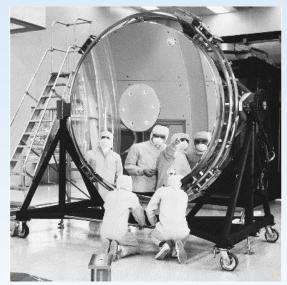
- Ceramic dielectric materials with engineered properties for microwave, millimeter wave communication system
- Dielectric ceramics as resonators, filters, oscilators
- Miniaturization continuing trend

Piezoceramic materials"

- Change in shape of reflector to improve signal quality
- Vibration control
- Positioning control



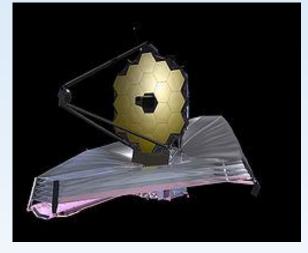
Use of Ceramics in Space Telescope Mirror



Hubble Space Telescope Glass mirror



Herschel Space Telescope SiC mirror



James Webb Telescope, electrostrictive ceramic actuator to control the shape of mirrors

Future requirements: Lower cost and increase in aerial density

Concluding Remarks

- Will see increasing use of CMCs in aircraft challenge to increase temperature capability to > 2700°F; cost reduction required
- Goal of Durable 3000°F CMC system for hypersonics and rocket propulsion still remains a major challenge
- Increasing use of piezoceramic and dielectric type of materials
 - Multifunctional structures will be future
 - Integration with components without adversely impacting component performance is challenging
 - Miniaturization will be the trend
- For high power density and high energy density systems, engineered porous materials through advanced manufacturing processes will be required
 - Additive manufacturing likely to play a role
 - Increasing use of nanomaterials
- Significant potential for improving ceramic materials for in-space propulsion